



CHANDRAYAAN 2

Exploring the Unknown



15 JULY, 2019



INTRODUCTION

Chandrayaan 2 is an Indian lunar mission that will boldly go where no country has ever gone before — the Moon's south polar region. We aim to improve our understanding of the Moon, which could lead to discoveries that will benefit India and humanity as a whole. These insights and experiences will cause a paradigm shift in how lunar expeditions are approached for years to come, propelling further voyages into the farthest frontiers.

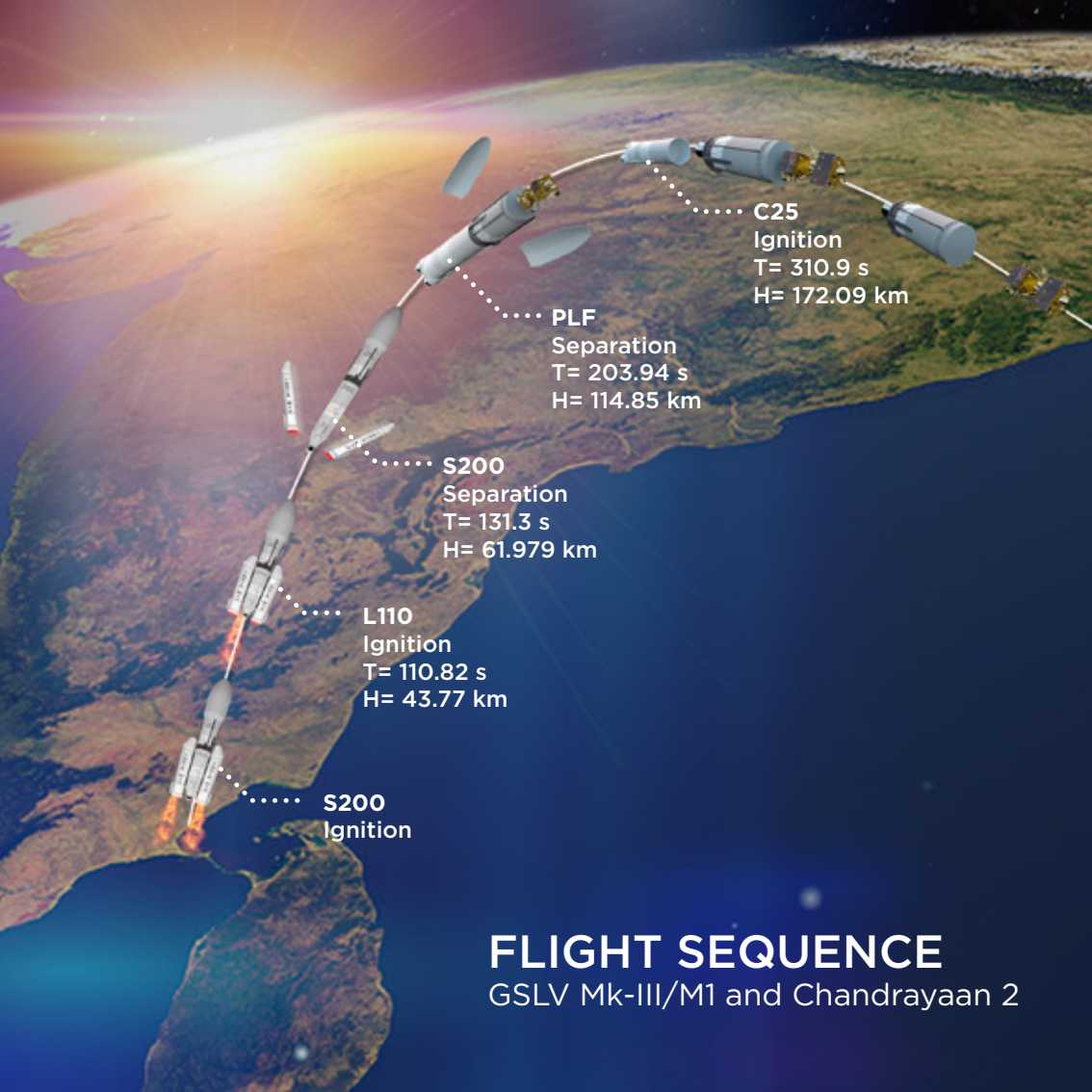
WHY ARE WE GOING TO THE MOON?

The Moon is the closest cosmic body on which space discovery can be attempted and documented.

It is also a promising test bed to demonstrate technologies required for deep-space missions.

Chandrayaan 2 attempts to foster a new age of discovery, increase our understanding of space, stimulate the advancement of technology, promote global alliances, and inspire a future generation of explorers and scientists.





..... S200
Ignition

..... L110
Ignition
T= 110.82 s
H= 43.77 km

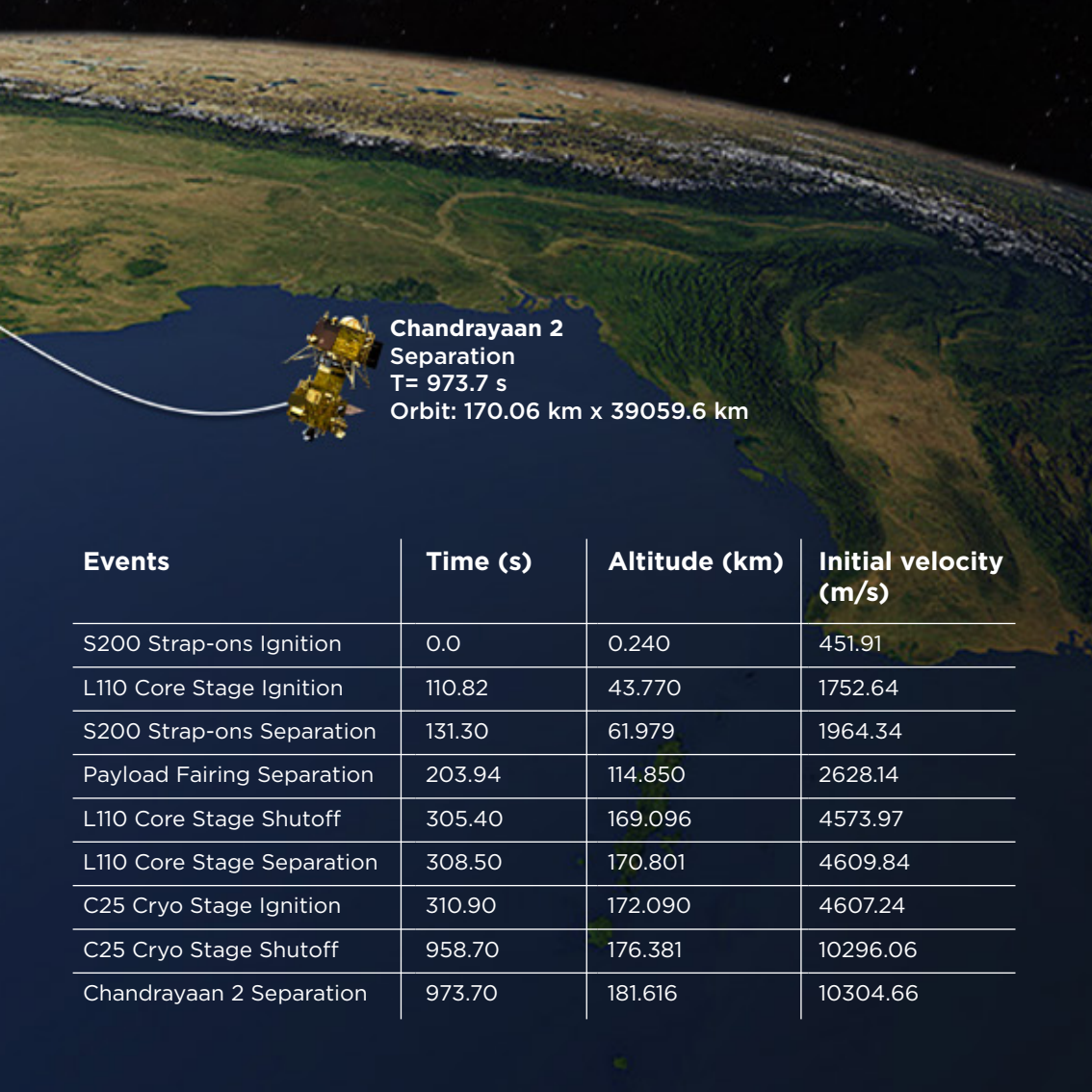
..... S200
Separation
T= 131.3 s
H= 61.979 km

..... PLF
Separation
T= 203.94 s
H= 114.85 km

..... C25
Ignition
T= 310.9 s
H= 172.09 km

FLIGHT SEQUENCE

GSLV Mk-III/M1 and Chandrayaan 2



Chandrayaan 2
Separation
T= 973.7 s
Orbit: 170.06 km x 39059.6 km

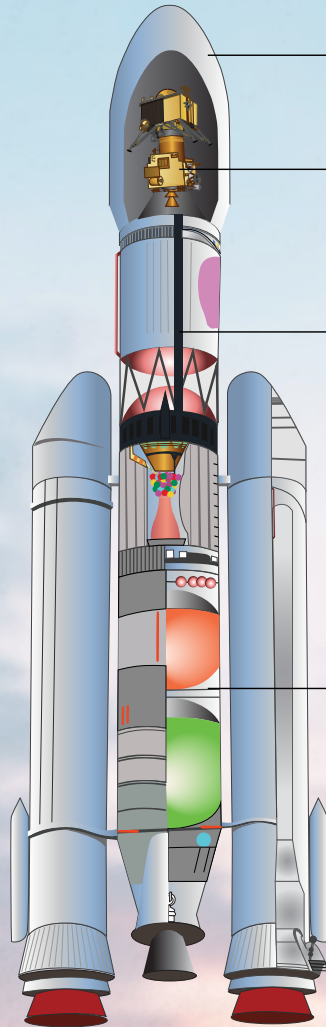
Events	Time (s)	Altitude (km)	Initial velocity (m/s)
S200 Strap-ons Ignition	0.0	0.240	451.91
L110 Core Stage Ignition	110.82	43.770	1752.64
S200 Strap-ons Separation	131.30	61.979	1964.34
Payload Fairing Separation	203.94	114.850	2628.14
L110 Core Stage Shutoff	305.40	169.096	4573.97
L110 Core Stage Separation	308.50	170.801	4609.84
C25 Cryo Stage Ignition	310.90	172.090	4607.24
C25 Cryo Stage Shutoff	958.70	176.381	10296.06
Chandrayaan 2 Separation	973.70	181.616	10304.66

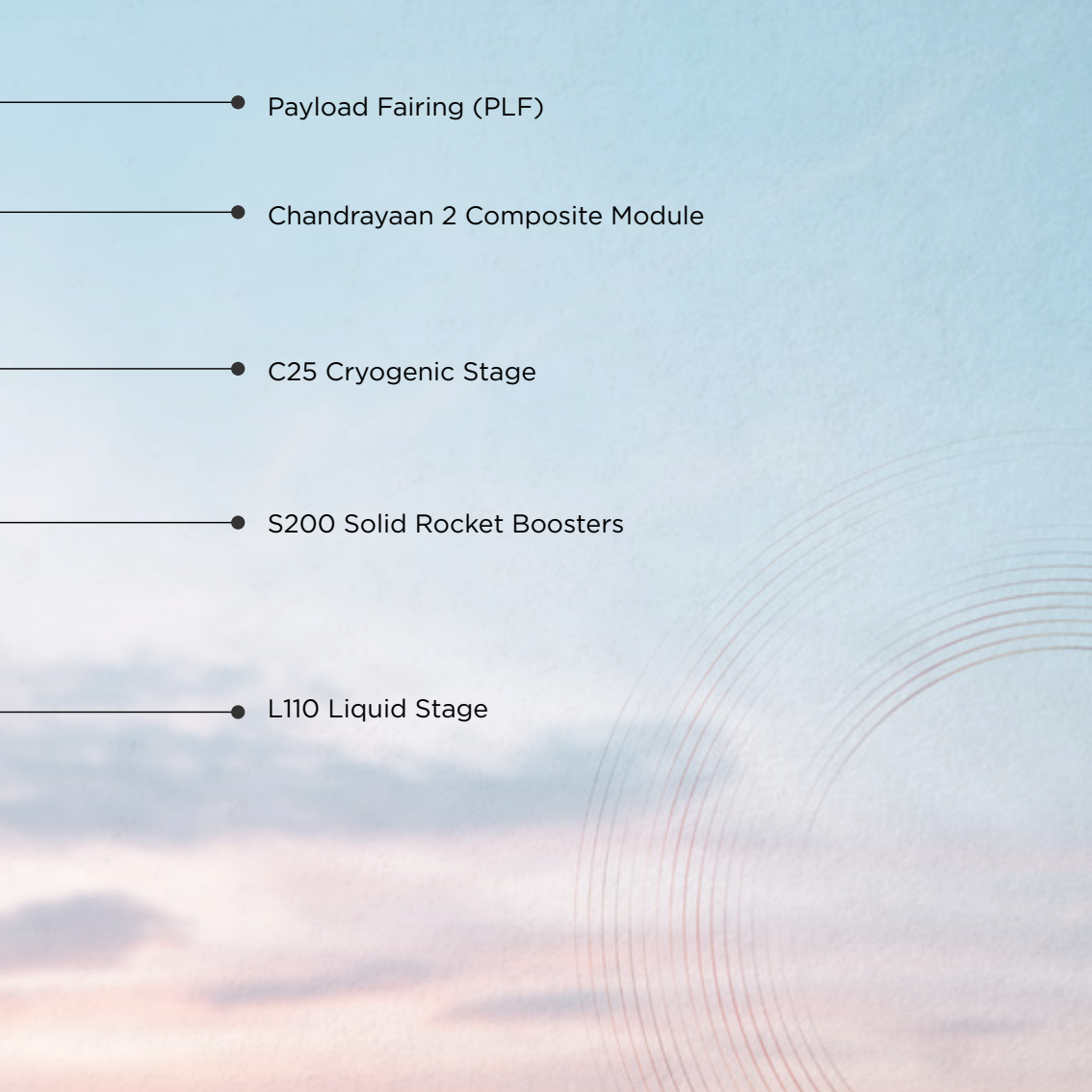
GEOSYNCHRONOUS SATELLITE LAUNCH VEHICLE MARK-III (GSLV Mk-III)

The GSLV Mk-III will carry Chandrayaan 2 to its designated orbit. **This three-stage vehicle is India's most powerful launcher to date**, and is capable of launching 4-tonne class of satellites to the Geosynchronous Transfer Orbit (GTO).

Height:
43.43 m

Lift Off Mass:
640 tonnes





- Payload Fairing (PLF)

- Chandrayaan 2 Composite Module

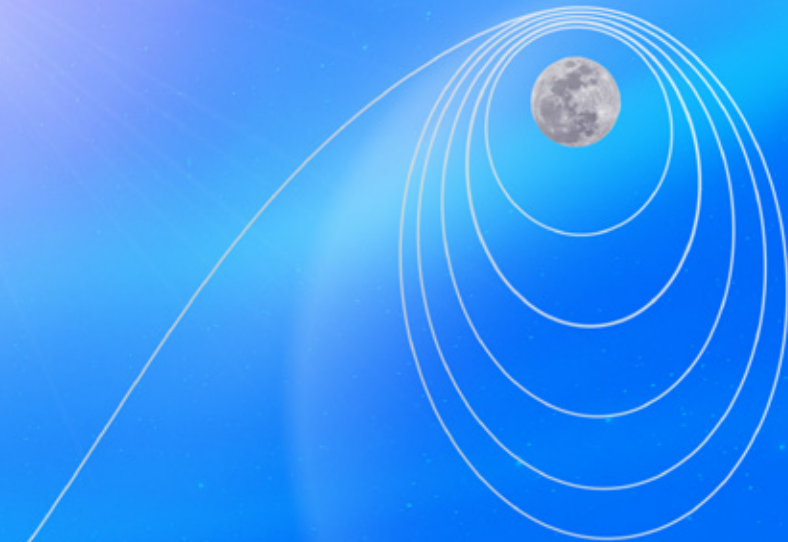
- C25 Cryogenic Stage

- S200 Solid Rocket Boosters

- L110 Liquid Stage

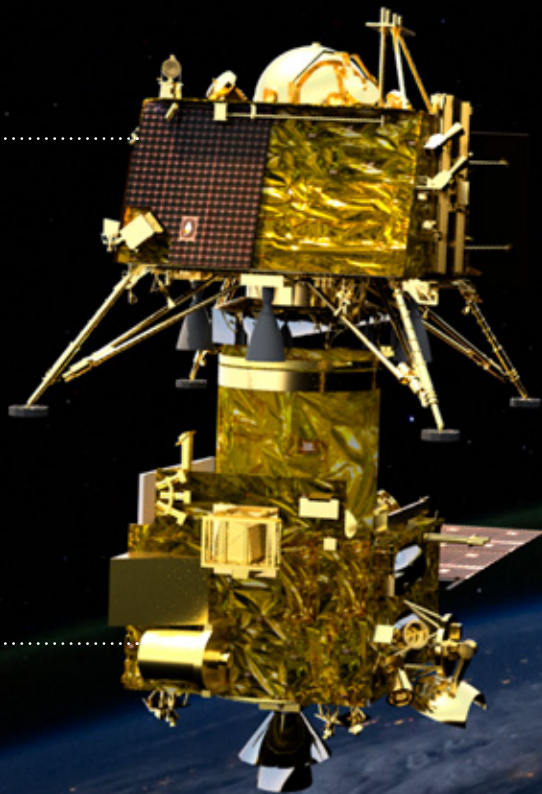
MISSION SEQUENCE





Phase	Period
Earth-bound Phase	Lift Off: 15 July Day 1 to Day 17 (17 days)
Trans Lunar Injection (TLI)	Day 17
Lunar Transfer Trajectory (LTT)	Day 17 to Day 22
Lunar Orbit Insertion (LOI)	Day 22
Lunar Bound Phase (LBN)	Day 22 to Day 49 (28 days)
Lander-Orbiter Separation	Day 50
Deboosting	Day 51
Powered Descent Starts	Day 54
Landing	Day 54

Vikram



Orbiter

CHANDRAYAAN 2

Composite Module

Chandrayaan 2 will be aided in achieving its mission by some of India's most advanced engineering marvels. Its composite module, which comprises technology and software developed across the country, includes a wholly indigenous rover and our first lander capable of executing a 'soft landing'.



Dimensions
3.1 x 3.1 x 5.8 m



Weight
3,850 kg

ORBITER



Weight
2,379 kg



Dimensions
3.2 x 5.8 x 2.1 m



Power
1,000 W

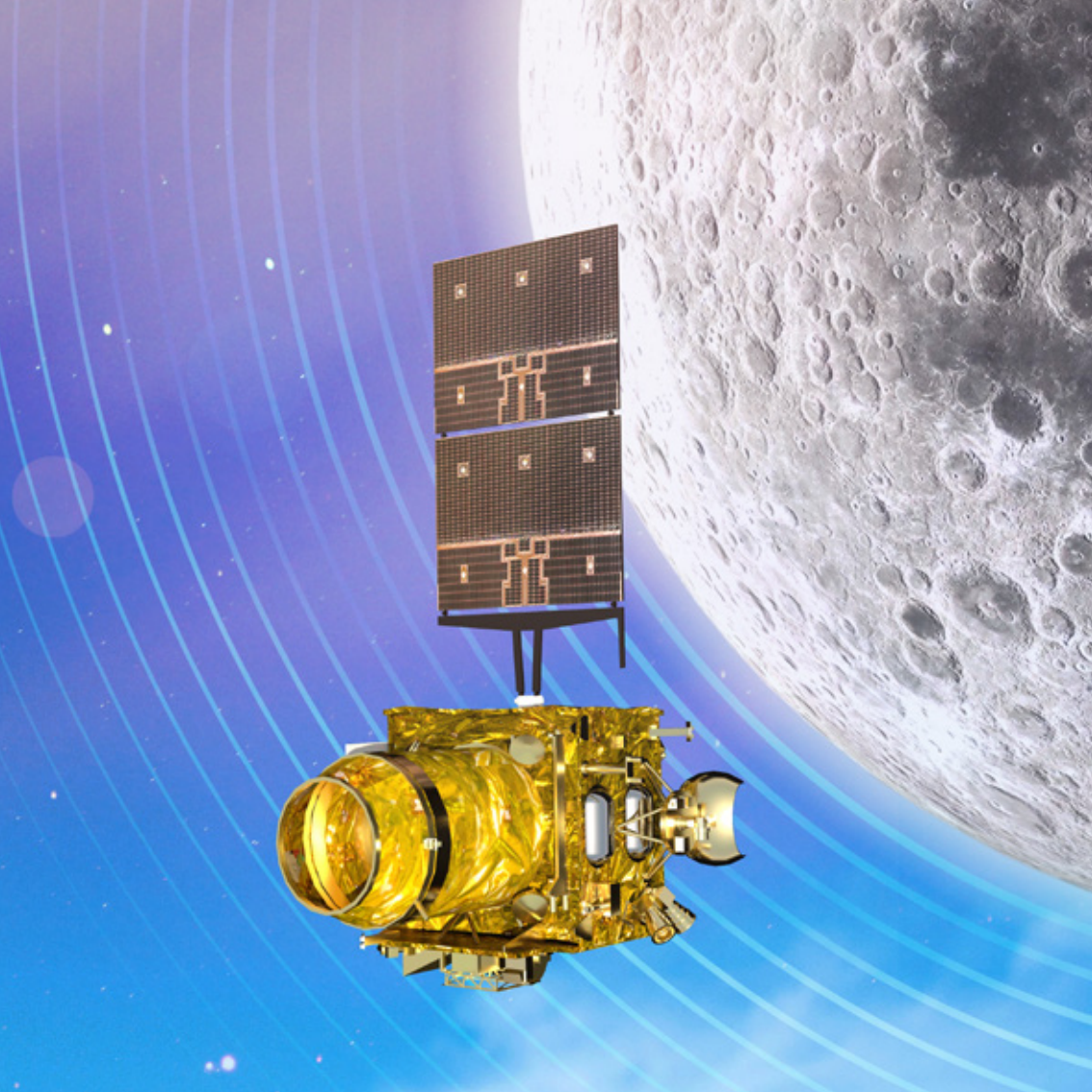


Payloads
8

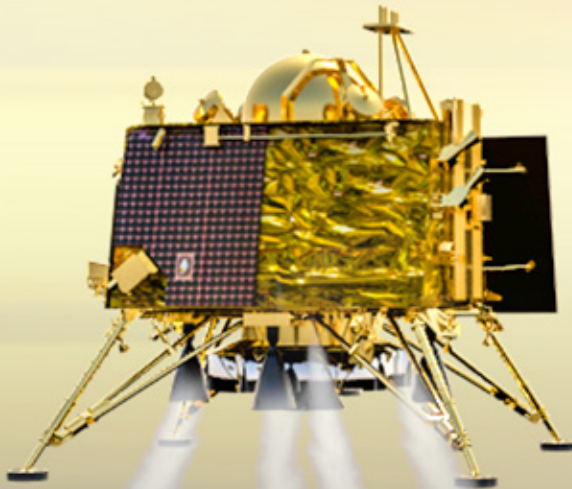


Mission Life
1 year in lunar orbit

At the time of launch, the Chandrayaan 2 **Orbiter will be capable of communicating with the Indian Deep Space Network (IDSN) at Byalalu, as well as with the Vikram lander.** The mission life of the Orbiter is one year, during which it will be placed in a 100 x 100 km lunar polar orbit.



VIKRAM LANDER





Weight
1,471 kg



Power
650 W



Payloads
4



Dimensions
**2.54 x 2 x
1.2 m**

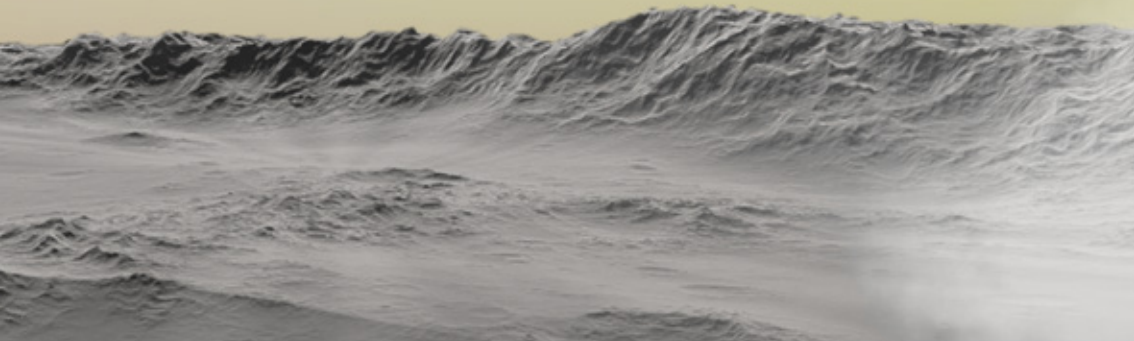


Mission Life
1 lunar day

Chandrayaan 2's lander is named Vikram after Dr Vikram A Sarabhai, the Father of the Indian Space Programme. **It is designed to function for one lunar day, which is equivalent to about 14 Earth days.** Vikram has the capability to communicate with IDSN at Byalalu near Bangalore, as well as with the Orbiter and Pragyan rover. The lander is designed to execute a soft landing on the lunar surface at a touchdown velocity of 2 metres per second.

Landing Site: High plain between two craters, Manzinus C and Simpelius N, at a latitude of about 70.9° South 22.7° East

Alternate Site: 67.7 ° South 18.4° West



PRAGYAN ROVER



Weight
27 kg



Power
50 W



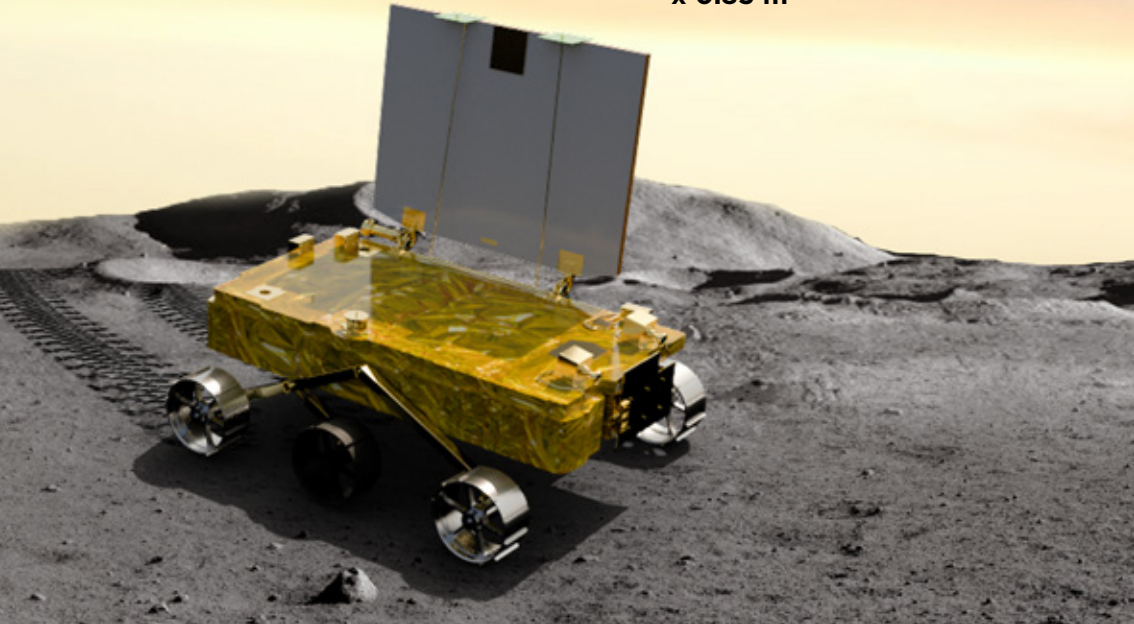
Payloads
2




Dimensions
**0.9 x 0.75
x 0.85 m**



Mission Life
1 lunar day

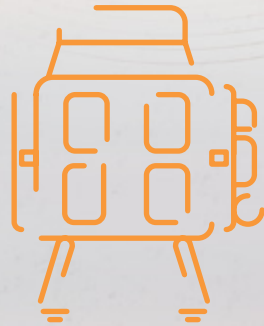




Chandrayaan 2's rover is a 6-wheeled robotic vehicle named Pragyan, which translates to 'wisdom' in Sanskrit. **It can travel up to 500 m (0.5 km) at a speed of 1 centimetre per second, and leverages solar energy for its functioning.** It can communicate with the lander.

MISSION PAYLOADS

Orbiter Payloads

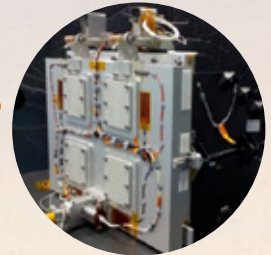


- **Terrain Mapping Camera:**
Will generate a Digital Elevation Model (DEM) of the entire Moon

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- **Chandrayaan 2 Large Area Soft X-ray Spectrometer:**
Will test the elemental composition of the Moon's surface

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- **Solar X-Ray Monitor:**
Will provide solar X-ray spectrum inputs for CLASS



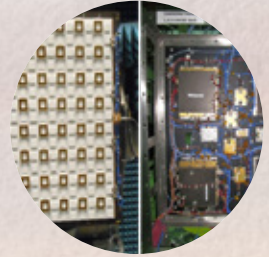
- **Imaging IR Spectrometer:**
Will map the Moon's mineralogy and confirm the presence of water-ice on the lunar surface

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● **Synthetic Aperture Radar L&S Bands:**

- Will map the polar region and confirm the presence of water-ice at the sub-surface level

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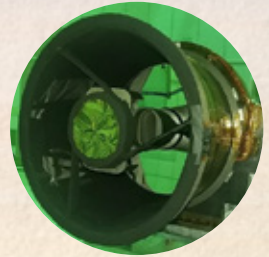
- **Chandra's Atmospheric Composition Explorer-2:**
Will examine the Moon's neutral exosphere

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● **Orbiter High Resolution Camera:**

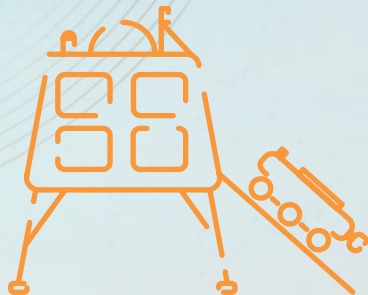
- Will conduct high-res topography mapping

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- **Dual Frequency Radio Science Experiment:**
Will study the lunar ionosphere

Vikram Payloads



- **Instrument for Lunar Seismic Activity:**

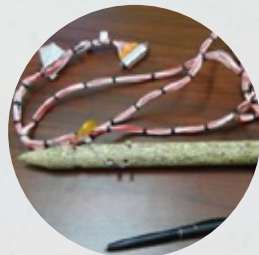
Will characterise the seismicity around the landing site

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- **Chandra's Surface Thermo-physical Experiment:**

Will examine the Moon's thermal conductivity and temperature gradient

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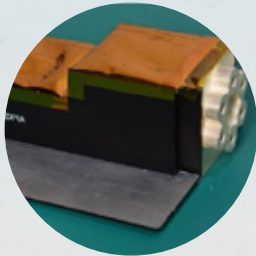
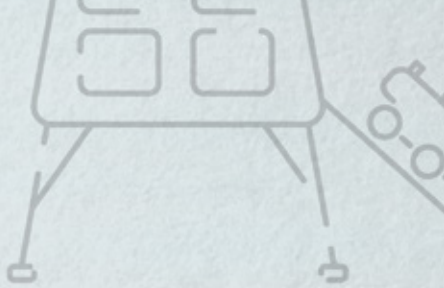


- **Langmuir Probe:**

Will conduct ionosphere studies on the lunar surface



Pragyan Payloads



- **Alpha Particle X-ray Spectrometer:**

Will determine the elemental composition of the Moon

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- **Laser Induced Breakdown Spectroscope:**

Will identify elemental abundance in the vicinity of the landing site



Passive Experiment

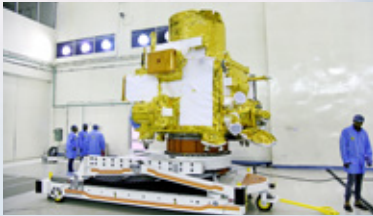
- **Laser Retroreflector Array (LRA):**

Will help us understand the dynamics of Earth's Moon system and also derive clues about the lunar interior





**BEHIND
THE SCENES**





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